

# NATIONAL UNIVERSITY OF SCIENCE AND TECHNOLOGY 

# FACULTY OF INDUSTRIAL TECHNOLOGY <br> DEPARTMENT OF CIVIL AND WATER ENGINEERING <br> FINITE ELEMENT METHOD IN CIVIL ENGINEERING 

TCW 5004

Examination Paper

December 2015

This examination paper consists of 7 pages
Time Allowed: 3 hours
Total Marks: 100
Special Requirements:
Examiner's Name: Miss Diana Makweche

## INSTRUCTIONS

1. Answer any four (4) questions
2. Each question carries 25 marks
3. Use of calculators is permissible

## MARK ALLOCATION

| QUESTION | MARKS |
| :--- | :--- |
| 1. | 25 |
| 2. | 25 |
| 3. | 25 |
| 4. | 25 |
| 5. | 25 |
| TOTAL | 100 |

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## QUESTION 1

(a) Using the principle of Minimum Potential Energy;
(i) Formulate the global stiffness matrix for the system in Figure Q1.
(ii) Solve for the unknown displacements
(iii) Calculate the unknown forces and reactions.


Figure Q1
(b) A bar of length $l$ is subjected to a uniformly distributed load of intensity $q$. The potential energy of the system can be expressed as

$$
\Pi(u(x))=U-W=\frac{1}{2} \int_{0}^{l} E A\left(\frac{d u}{d x}\right)^{2} d x-\int_{0}^{l} q u d x
$$

Using the trial function $\mathrm{u}(x)=a x$,
(i) Determine the coefficient $a$
(ii) Find expressions for strain, $\varepsilon$, and stress, $\sigma$. [15]

## QUESTION 2

(a) The bar element has two nodes and the linear displacement function is as shown in Figure Q2A.
(i) What are the equations for linear interpolation (shape) functions $N_{1}$ and $N_{2}$ ?
(ii) Sketch $N_{1}$ and $N_{2}$
(iii) What is the purpose of a shape function?


Figure Q2A
(b) (i) Explain how you would apply a fixed support to node 1 of the beam shown in Figure Q2B.
(ii) Reduce the uniformly distributed load into the equivalent nodal forces and moments. Give the answer in the form of the global force vector $\mathbf{F}$.


Figure Q2B

## QUESTION 3

The tapered bar in Figure Q3 has a linearly varying cross-sectional area which varies from $3 \mathrm{~A}_{0}$ at the left side to $\mathrm{A}_{0}$ at the right side. The bar is divided into two elements. Show that the stiffness matrix for the system is $\left[\begin{array}{ccc}\frac{3 k}{2} & \frac{-3 k}{2} & 0 \\ \frac{-3 k}{2} & 4 k & \frac{-5 k}{2} \\ 0 & \frac{-5 k}{2} & \frac{5 k}{2}\end{array}\right] \quad$ here $k=\frac{E A_{0}}{L}$.

Find a solution by integrating $\sigma_{x}=E\left(\frac{d u}{d x}\right)=\frac{P}{A}(x)$ subject to the boundary condition $u(0)=0$


Figure Q3

## QUESTION 4

(a) What is the difference between a beam and a bar (rod) element.
(b) The truss in Figure Q4 is composed of two members each of cross-sectional area $968 \mathrm{~mm} 2 . \mathrm{E}=70 \mathrm{GPa}$.
(i) Construct the individual elemental matrices
(ii) Assemble the global matrix


Figure Q4

## QUESTION 5

The structure in Figure Q5 is subjected only to gravitational forces acting on the elements.


Figure Q5
(i) Derive the stiffness matrix for each element and assemble the global stiffness matrix
(ii) Construct the force vector considering the gravitational force acting on the elements
(iii) Apply the relevant boundary conditions and solve for the nodal displacements
(iv) Calculate the reaction force at the support, element strains and element stresses.

## FORMULAE

$\left\{\begin{array}{l}\hat{f}_{1 x} \\ \hat{f}_{2 x}\end{array}\right\}=\frac{A E}{L}\left[\begin{array}{rr}1 & -1 \\ -1 & 1\end{array}\right]\left\{\begin{array}{l}\hat{d}_{1 x} \\ \hat{d}_{2 x}\end{array}\right\}$

$$
\left\{\begin{array}{l}
\hat{d}_{x} \\
\hat{d}_{y}
\end{array}\right\}=\left[\begin{array}{rr}
C & S \\
-S & C
\end{array}\right]\left\{\begin{array}{l}
d_{x} \\
d_{y}
\end{array}\right]
$$

$$
\underline{k}=\frac{A E}{L}\left[\begin{array}{cccc}
C^{2} & C S & -C^{2} & -C S \\
& S^{2} & -C S & -S^{2} \\
& & C^{2} & C S \\
\text { Symmetry } & & S^{2}
\end{array}\right]
$$

$$
\left\{\begin{array}{l}
\hat{f}_{y_{y}} \\
m_{1} \\
\hat{f}_{2} \\
\hat{m}_{2}
\end{array}\right\}=\frac{E I}{L^{3}}\left[\begin{array}{cccc}
12 & 6 L & -12 & 6 L \\
6 L & 4 L^{2} & -6 L & 2 L^{2} \\
-12 & -6 L & 12 & -6 L \\
6 L & 2 L^{2} & -6 L & 4 L^{2}
\end{array}\right]\left\{\begin{array}{c}
\hat{d}_{1 y} \\
\hat{\phi}_{1} \\
\hat{d}_{2 y} \\
\hat{\phi}_{2}
\end{array}\right\}
$$



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